



P.E.S. College of Engineering, Mandya - 571 401

(An Autonomous Institution affiliated to VTU, Belagavi)

Third Semester, B.E. - Automobile Engineering

Semester End Examination; March/April - 2022

Fluid Mechanics

Time: 3 hrs

Max. Marks: 100

Course Outcome

The Students will be able to:

CO1: Understand and Explain various properties of fluids, Fluid - statics, kinematics & Dynamics and the basic concepts of Fluid mechanics

CO2: Apply, Interpret and describe about laminar flow, compressible flow, Energy Losses in Flow through pipes and dimensional analysis about various primary & secondary units.

CO3: Derive Equations for fluids properties, Fluid - statics, kinematics & Dynamics and their applications.

CO4: Analyze/Compare, solve engineering problems involving fluid flow pertaining to fluids properties, Fluid - statics, kinematics & Dynamics and their applications.

CO5: Analyze and solve engineering problems pertaining fluid flow losses, dimensional analysis techniques and practical applications of fluid mechanics in compressible flow.

Note: I) PART - A is compulsory. Two marks for each question.

II) PART - B: Answer any **Two** sub questions (from a, b, c) for Maximum of **18 marks** from each unit.

Q. No.	Questions	Marks	BLs	COs	POs
I : PART - A		10			
I a.	State Newton's law of viscosity.	2	L1	CO1	PO1
b.	What is pressure head?	2	L1	CO2	PO1
c.	What is the condition for stable equilibrium for floating body?	2	L3	CO3	PO1
d.	Write the Bernoulli's equation for real fluid.	2	L2	CO4	PO1
e.	Explain terminal velocity of a body.	2	L2	CO5	PO1
II : PART - B		90			
UNIT - I		18			
1 a.	State and prove Hydrostatic law using rectangular parallel piped element.	12	L2	CO1	PO1
b.	A simple U-tube manometer containing mercury is connected to a pipe in which a fluid of specific gravity 0.8 and having vacuum pressure is flowing. The other end of the manometer is open to atmosphere. Find the vacuum pressure in pipe, if the difference of mercury level in the two limbs is 40 cm and the height of fluid in the lift from the centre of pipe is 15 cm below.	12	L3	CO1	PO2
c.	Distinguish between absolute pressure, Gauge pressure and vacuum pressure.	6	L2	CO1	PO1

UNIT - II**18**

- 2 a. A circular plate 1.5 m diameter is submerged in water with its greatest and least depths below the surface being 2 m and 0.75 m respectively. Determine;
- 12 L3 CO2 PO2
- i) The total pressure on the face of the plate
- ii) The position of the centre of pressure
- b. Determine the meta centric height by analytical method. 12 L3 CO2 PO1
- c. Define;
- i) Total pressure 6 L1 CO2 PO1
- ii) Centre of pressure

UNIT - III**18**

- 3 a. Define the equation of continuity. Obtain an expression for the continuity equation for a three-dimensional flow. 12 L2 CO3 PO1
- b. A pipe line carrying oil of specific gravity 0.9 changes in diameter from 0.2 m at a position A to 0.4 m diameter at a position B, which is 3.5 m at a higher level. If the pressure at A and B are 98,100 N/m² and 58,860 N/m² respectively and the discharge is 0.2 m³. Determine the loss of head and direction of flow. 12 L3 CO3 PO2
- c. With neat sketch, explain pitot tube to measure the velocity of a flowing liquid. 6 L2 CO3 PO1

UNIT - IV**18**

- 4 a. An Oil with density 800 kg/m³ and viscosity 0.16 N-s/m² flows through a 20 cm diameter pipe. The loss of head due to fluid friction over a 100 m length of pipe is 1.3 m of oil. Determine;
- 12 L3 CO4 PO2
- i) The average velocity of the flow ii) The volumetric flow rate
- iii) The wall shear stress iv) The Darcy's friction factor
- b. Show that the velocity of sound wave in compressible fluid is given by,
- 12 L2 CO4 PO1
- $$C = \sqrt{E/\rho}$$
- c. Explain the terms mach number mach cone and mach angle. 6 L2 CO4 PO2

UNIT - V**18**

- 5 a. Two reservoirs are connected by a pipeline of diameter 30 cm and length 600 m. If the difference of water surface in the reservoir is 8 m, find the rate of flow. Take Darcy's friction factor $f = 0.03$. Consider only the loss due to fluid friction. 12 L3 CO5 PO2

- b. Drag force F_D on a high speed aircraft depends on the velocity of flight V . The characteristic geometrical dimension of the aircraft L_1 , the density S , the viscosity μ and isentropic bulk modulus of elasticity E_s of the ambient air. Using Buckingham's Π theorem. Find out the independent dimensionless quantities, which describe the phenomenon drag on the aircraft. 12 L3 CO5 PO2
- c. What is hydraulic gradient line and total energy line? 6 L1 CO5 PO1

* * *